MONITORING INSTRUCTION FOR INSPECTION

Issue No. 01 Rev. No. 02

Date of Issue 17.06.21

ALUMINIUM ALLOY BAR Ø75mm

HAPP/QA/SC/C/014

Rev no	Amendment	Date
2	Material spec amended as per Document No.: DRDO-ARDE-DOA-TOT-209-2017 Rev No:00 Dtd: May 2021	17.06.21

MATERIAL SPECIFICATION

Aluminum Alloy to specification DTD 5124, or Specification

ASTM B 221M, Grade 7075-T6510, or HAPP/QA/SPEC/008

DOCUMENT END USE : DRDO-ARDE-DOA-TOT-209-2017 Rev No: 00 Dtd: May 2021

: Tail unit 120/125 mm MK-I FSAPDS & 125mm Practice Shot

TABLE A. INSPECTION CHECK TO BE CARRIED OUT AT FIRM'S PREMISES BY THE FIRM & ON RECIEPT AT HEPF

SL. NO.	CHARACTERITICS	SPECIFICATION / REQUIREMENT	SAMPLE SIZE
1	Workmanship (Visual)	The material shall be uniform and free from lamination, pipes, silvers, laps, cracks, kinks, warps, wrinkles, deep scratches and other injurious defects.	400%
2	Dimensions	Outer Diameter = 75.0 ± 0.5mm ,Length = 3360mm(min), 5580mm (max). In Integral Multiples of 111mm Having Tolerance ± 5mm	100%
3	Chemical Composition (%)	Aluminum Alloy to specification DTD 5124, or Specification ASTM B 221M, Grade 7075-T6510 or HAPP/QA/SPEC/008	One sample for each heat / cast
4	Mechanical Test	UTS – 540 MPa(min) 0.2% PS - 470 MPa (min) % Elongation – 6% Min.	2 samples for each heat / cast
5	*Ultrasonic Testing	ASTM B 594 Type 1 Class A	
6	Peripheral Grain Growth	Limited upto 1.25mm on max diameter	100%
7	Marking	Each bar shall be legibly stamped with manufacturer's identity & heat number.	Each
8	Packing	Packing of the material shall be done in such a manner to avoid damage in handling and transit.	Consignment

NOTE:

- THE RAW MATERIAL TO BE TESTED BY THE FIRM BY SELECTING THE SAMPLE BY THE FIRM ITSELF-AND FORWARDED TO NEAR BY NABL ACCREDITED LABORATORY OR GOVT.APPROVED LAB FOR TESTING,
 - *ULTRASONIC TESTING TO BE CARRIED OUT AT FIRMS PREMISES BY THE FIRM.
- 2. THE FIRM HAS TO CHECK FOR THE DIMENSIONS, VISUAL DEFECTS AND MARKING AS PER TABLE A. AFTER COMPLETION OF DIMENSIONAL, CHEMICAL AND MECHANICAL PROPERTIES. THE FIRM HAS TO FORWARD THE FOLLOWING DOCUMENTS TO HEPF/TRICHY.
 - THE RAW MATERIAL ORIGINAL MANUFACTURER'S CERTIFICATE, HEAT NUMBER, QUANTITY PURCHASED AND NUMBER OF BARS ARE TO BE MENTIONED IN THE INSPECTION LETTER TO HEPF/TRICHY.
 - II. THE CHEMICAL AND MECHANICAL TEST CERTIFICATES FROM NABL ACCREDITED LAB OR GOVT APPROVED LAB AS PER TABLE A.
 - III. DIMENSIONAL REPORT INCLUDING VISUAL AS PER TABLE A.
- 3 . ALL THE ABOVE DOCUMENTS MENTIONED AT NOTE NO.2 ABOVE ARE TO BE FORWARDED TO GM/HAPP.
- 4. MATERIAL HAS TO BE DESPATCHED TO HEPF ONLY AFTER DUE SUBMISSION OF ALL RELATED DOCUMENTS AS DETAILED IN TABLE A AND AFTER GETTING CLEARANCE FROM HEPF / TRICHY.

MONITORING INSTRUCTION FOR INSPECTION

Issue No. 01 Rev. No. 02

Date of Issue 17.06.21

ALUMINIUM ALLOY BAR Ø75mm

HAPP/QA/SC/C/014

Rev no	Amendment	Date
2	Material spec amended as per Document No.: DRDO-ARDE-DOA-TOT-209- 2017 Rev No:00 Dtd: May 2021	17.06.21

TABLE B. VERIFICATION OF INSPECTION DOCUMENTS.

FOLLOWING INSPECTION DOCUMENTS MUST BE ENCLOSED WITH EACH SUPPLY

	THE ENGLOSED WITH EACH SUPPLY.
SL. NO.	INSPECTION DOCUMENTS
1	THE RAW MATERIAL ORIGINAL MANUFACTURER'S CERTIFICATE, DETAILS OF HEAT NUMBER, QUANTITY PURCHASED AND NUMBER OF BARS ETC.
2	THE CHEMICAL AND MECHANICAL TEST CERTIFICATES FROM NABL ACCREDITED OR GOVT APPROVED LAB .
3	DIMENSION REPORT INCLUDING VISUAL
4	PACKING SLIP DETAILS
5	IN ADDITION TO THE ABOVE SOFT COPIES OF ALL THE CERTIFICATES MENTIONED IN TABLE - A SHALL BE SENT TO E-MAIL ID's. happqa.ofb@ofb.gov.in , mmhapp.ofb@ofb.gov.in
NOTE	1.IN CASE OF ANY DIFFERENCES, QAP & DRAWING SHALL PREVAIL 2. EXPLICIT DEVIATION(S) IF ANY SUCH AS TYPOGRAPHICAL ERROR, VALUES, NUMERIC, OTHER PARAMETER, ETC IS/ARE FOUND IN MONITORING INSTRUCTION OF THE ABOVE STORES, THE RELEVANT STANDARD CONFORMING TO THE SPECIFICATION SHALL BE REFERRED TO CONFIRM THE PARAMETER

SREENIVASA RAO BODALA DGM / PM MEMBER / MI COMMITTEE ANHILESH GAUR WM / P

MEMBER / MI COMMITTEE

SEINUNNOM JOSEPH KIPGEN

DGM / QA

MEMBER /MI COMMITTEE

T.PRABHU JT.GM / QA
CHAIRMAN / MI COMMITTEE

ISSUED BY
STANDARD CELL
MP cheet NR. 240 454
DATE: 3/3/27 SIGNATURE:

TECHNICAL SPECIFICATION

FOR ALUMINIUM ALLOY 7075-T6510 EXTRUDED BARS TO ASTM B 221 M

No. HAPP/QA/SPEC/008

Revision No.: **02** Revn. Date: 01-12-05

1. INSPECTION AND TESTING PROCEDURE

This specification shall be used in conjunction with the relevant sections of ASTM STANDARD B 221 M-96. Alloy 7075 T 6510.

2. QUALITY OF MATERIAL

The material shall be made from Aluminium and alloying constituents, with or without approved scrap, at the discretion of the manufacturer.

Preference: Material Shall be Single Strand Extrusion.

3. CHEMICAL COMPOSITION

The chemical composition of the material shall be :-

Elements	Perce	ntage
	Min	Max
Copper	1.2	2.0
Magnesium	2.1	2.9
Silicon	-	0.4
Iron	-	0.5
Manganese	-	0.3
Zinc	5.1	6.1
Titanium	-	0.2
Chromium	0.18	0.28
Other Elements*	Each 0.05 max	Total 0.15 max
Aluminium	The ren	nainder

*Subject to the discretion of the Inspection Authority, determination of these elements need be made on a small proportion only of the samples analysed.

4. <u>CONDITION</u>

4.1 BARS AND EXTRUDED SECTIONS

Unless otherwise agreed and stated on the order, bars for machining and extruded sections shall be supplied in solution treated, stress relieved (by control stretching to a permanent extension not less than 1.5 % and not more than 2.5 %) and precipitation treated condition to T 6510.

The material shall be uniform in quality and shall be free from lamination pipes, slivers, cracks, kinks, warps, wrinkles, deep scratches, and other injurious defects. However, minor marks on the surface of the bars due to heat treatment in the furnaces, minor spots which have been ground in preparation for the ultrasonic testing is acceptable. Material will be supplied having Diameter 135,00 ± 0.50 mm.

R. PARTH SARAFHY SIGNION

TABLE 1 Chemical Composition Limits A.B.C

Alfoy	Silicon	Iron	Соррег	Manga-	Magne-	Chromium	Zinc	Titanium	Vanadium	Other	Elements ^D	A Learning upo
1000				nese	sium			ritanium	variatium	Each	Total ^E	— AJuminum
1060 1100	0.25	0.35	0.05	0.03	0.03	5019	0.05	0.03	0.05	0.03		99,60 min ^F
2014	0.95	Si + Fe	0.05-0.20	0.05	(0.6)	3960	0.10	444	***	0.05	0.15	99.00 min F
2024		0.7	3.9-5.0	0.40-1,2	0.20 - 0.8	0.10	0.25	0.15 ^G	237	0.05 ^G	0.15	remainder
2219	0.50	0.50	3.8-4.9	0.30-0.9	1.2-1,8	0.10	0.25	0.15 ^a	22/27	0.050	0.15	remainder
3003	0.20	0.30	5.8-6.8	0.20-0.40	0.02	627	0.10	0.02-0.10	0.05-0.15	0.05 ^H	0.15	remainder
Alclad 3003	0.6	0.7	0.05-0.20	1.0-1.5	***	222)	0.10	.1777	***	0.05	0.15	remainder
3004	885		Clad with 70		200	2221	***	11111	344	:706	446	Way
3102	0.30	0.7	0.25	1.0-1.5	0.8 - 1.3	440	0.25	799	344 C	0.05	0.15	rernainder
	0.40	0.7	0.10	0.05 - 0.40	100	***	0.30	0.10	198	0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2-2.8	0.15-0.35	0.10	7997	777	0.05	0.15	remainder
5083	0.40	0.40	0.10	0.40-1.0	4.0-4.9	0.05-0.25	0.25	0.15		0.05	0.15	remainder
5086	0.40	0.50	0.10	0.20 - 0.7	3.5-4.5	0.05-0.25	0.25	0.15		0.05	0.15	remainder
5154	0.25	0.40	0.10	0.10	3.1-3.9	0.15-0.35	0.20	0.20	***	0.05	0.15	remainder
5454	0.25	0.40	0,10	0.50 - 1.0	2.4-3.0	0.05-0.20	0.25	0.20	5500	0.05		remainder
5456	0.25	0.40	0.10	0.50-1.0	4.7-5.5	0.05-0.20	0.25	0.20	***	0.05	0.15	
6005	0.6-0.9	0.35	0.10	0.10	0.40-0.6	0.10	0.10	0.10	112.2		0.15	remainder
6005A	0.50-0.9	0.35	0.30	0.50'	0.40-0.7	0.30'	0.20	0.10	10:-	0.05	0.15	remainder
6060	0.30-0.6	0.10-0.30	0.10	0.10	0.35-0.6	0.5	0.15	0.10	44.0	0.05	0.15	remainder
6061 ⁷	0.40-0.8	0.7	0.15-0.40	0.15	0.8-1.2	0.04-0.35	0.15	0.15	100	0.05	0.15	rennalnder
6063	0.20-0.6	0.35	0.10	0.10	0.45-0.9	0.10	0.10	0.15	lear-	0.05	0.15	remainder
6066	0.9-1.8	0.50	0.7-1.2	0.6-1.1	0.8-1.4	0.40	0.10		***	0.05	0.15	remainder
6070	1.0-1.7	0.50	0.15-0.40	0.401.0	0.50-1.2	0.10		0.20	200	0.05	0.15	remainder
6082	0.7-1.3	0.50	0.10	0.40-1.0	0.6-1.2	0.10	0.25	0.15	144	0.05	0.15	remainder
6105	0.6-1.0	0.35	0.10	0.15	0.45-0.8	0.10	0.20	0.10	597	0.05	0.15	remainder
6162	0.40-0.8	0.50	0.20	0.10			0.10	0.10	200	0.05	0.15	remainder
6262	0.40-0.8	0.7	0.15-0.40	0.15	0.71.1	0.10	0.25	0.10	3.649	0.05	0.15	romainder
6351	0.7-1.3	0.50	0.10	0.15	0.8-1.2	0.04-0.14	0.25	0.15	-144	0.05 ^K	0.15 ^K	remainder
6463	0.20-0.6	0.15	0.10	0.40-0.8	0.40-0.8	\$500	0.20	0.20	CHÁ	0.05	0.15	remainder
6560	0.30-0.7	0.10-0.30	0.20		0.45-0.9	369	0.05	94.40	444	0.05	0.15	remainder
7005	0.35	0.10-0.30	0.05-0.20	0.20	0.20-0.6	0.05	0.15	0.10		0.05	0.15	remainder
7072 ^M		SI + Fe	0.10	0.20-0.7	1.0-1.8	0.06-0.20	4.0-5.0	0.01-0.06		0.05^{L}	0.15 ^L	remainder
7075	0.40	0.50		0.10	0.10	300	0.8-1.3	***	0550			remainder
7116	0.15	0.30	1.2-2.0	0.30	2.1-2.9	0.18-0.28	5.1-6.1	0.20 ^N	1918	0.05^	0.15	remainder
7129	0.15	0.30	0.50-1.1	0.05	0.9-1.4	744	4.2-5.2	0.05	0.05	0.050	0.15	remainder
7178	0.15	0.30	0.50-0.9	0.10	1.3-2.0	0.10	4.2-5.2	0.05	0.05	0.050	0.15	remainder
	0.40	0.50	1.6-2.4	0.30	2.4-3.1	0.18-0,28	6.3-7.3	0.20	944	0.05	0.15	remainder

 4 Limits are in weight percent maximum unless shown as a range, or stated otherwise.

Analysis shall be made for the elements for which limits are shown in this table, © For the purpose of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in

the last right-hand place of the figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E29.

Others includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered

FOlher Elements—Total shall be the sum of unspecified metallic elements 0.010 % or more, rounded to the second decimal before determining the sum.

Fine aluminum content shall be calculated by subtracting from 100.00 % the sum of all metallic elements present in amounts of 0.010 % or more each, rounded to the

second decimal before determining the sum.

a Upon agreement between the purchaser and the producer or supplier, a Zr + TI limit of 0.20 % max is permitted. Properties in Specification (Table 2) are not based on the Zirconium and Titanium algorithm.

H Zirconium, 0.10-0.25 %. The total for other elements does not include zirconium.

'Manganese plus chromium shall total 0.12-0.50.

In 1985 the requirements for 6062 were combined with those for 6061 by revising the minimum chromium from "0.15 %" to" 0.04 %." This action cancelled alloy 6062.

K Bismuth and lead shall be 0.40-0.7 % each.

La Zirconium 0.08-0.20 %. The total for other elements does not include zirconium.

M Composition of cladding alloy applied during the course of manufacture. Samples from finished tubo shall not be required to conform to these limits. Nupon agreement between the purchaser and the producor or supplier, a Zr + Tl limit of 0.25 % max is permitted. Proporties in Specification (Table 2) are no⁴ based

on the Zirconium and Titanium algorithm.
O Gallium 0.03 % max.

E607 Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point to Plane Technique Nitrogen Atmosphere

E716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis

E1004 Test Method for Determining Electrical Conductivity Using the Electromagnetic (Eddy-Current) Method

E1251 Test Method for Analysis of Aluminum and Aluminum Alloys by Atomic Emission Spectrometry

G47 Test Method for Determining Susceptibility to Stress-

Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy

Method of Test for Exfoliation Corrosion Susceptibil ity in 7XXX Series Copper-Containing Aluminum Alloys (EXCO Test) (G34-72)3

2.3 ANSI Standards:2

^{*}The applicable edition in the use of this specification is G34-72, we with is available in the gray pages of the Annual Book of ASTM Standards, Vol (2 - O2.

RESTRICTED

Technical Specification & Quality Document			Document No.: DRDO-ARDE-DOA-TOT-209-2017						
Issue:01	Date	May 2021	Revision No.	00	Date:	May 2021	Page 24		
Component / Assembly			125mm FSAPDS Practice Shot Assembly (Empty)						

TAIL UNIT

1. Drawing No.

: 9607 01 03 01 03 001 00TC

2. Method of manufacture

: Machining and Hard Anodizing

3. Receiving Inspection

3.1 Raw material

: Aluminum Alloy to specification DTD 5124, or

Specification ASTM B 221M, Grade 7075-T6510, or

HAPP/QA/SPEC/008

Tests/checks and acceptance criteria for raw material:

1. Chemical analysis as per specification DTD 5124 or ASTM B 221M, Grade 7075-T6510 or HAPP/QA/SPEC/008

2. Mechanical properties:

Sr. No.	Properties	Acceptance criteria
1,	UTS	540 MPa (min)
2.	0.2 % Proof Stress	470 MPa (min)
3.	% Elongation	6% (min)

3. Metallographic Inspection:

1,	Peripheral Grain	Limited up to 1.25mm on max Dia
	Growth	20

4. In-process inspection

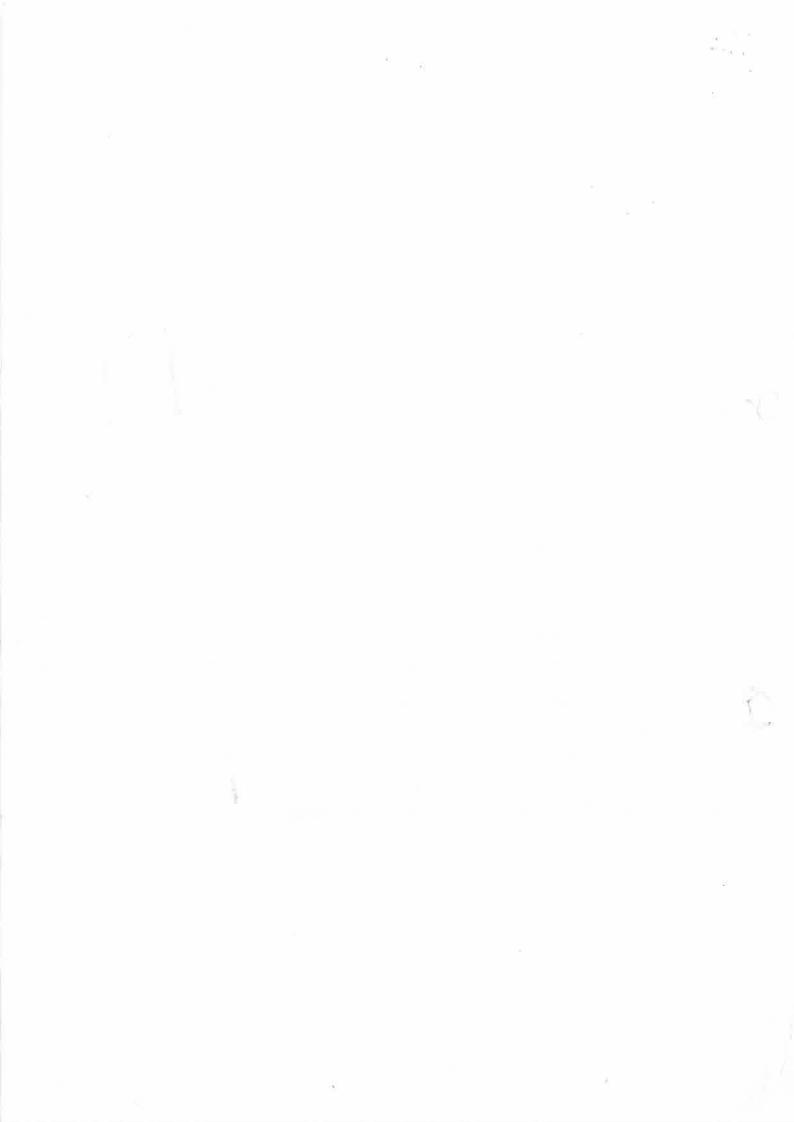
: Nil

5. Stage inspection

: Nil

- 6. Final Inspection:
- **6.1.** Visual inspection:
- 6.1.1 Features for visual examination and acceptance criteria:

Sr. No.	Details of features	Sample size	Acceptance criteria	
1.	Burrs	100 %	Not permitted	
2.	Damage to Thread	100 %	Not permitted	
3.	Poor surface finish	100 %	Not permitted	





Designation: B221M - 07

Standard Specification for Aluminum and Aluminum-Alloy Extruded Bars, Rods, Wire, Profiles, and Tubes (Metric)¹

This standard is issued under the fixed designation B221M; the number immediately following the designation indicates the year of original adoption or, in the case of revision, the year of last revision. A number in parentheses indicates the year of last reapproval. A superscript epsilon (ε) indicates an editorial change since the last revision or reapproval.

This standard has been approved for use by agencies of the Department of Defense,

Note-Note 3 added and Sections 2.2, 9, X2.1, and Table 2 updated and the year date was changed on Aug. 10, 2007

1. Scope*

1.1 This specification covers aluminum and aluminum-alloy extruded bar, rod, wire, profile, and tube in the aluminum alloys (Note 1) and tempers shown in Table 2.

Nore 1—Throughout this specification the use of the term alloy in the general sense includes aluminum as well as aluminum alloy.

Note 2—For rolled or cold-finished bars and rods refer to Specification B211M, for drawn seamless tube used in pressure applications, Specification B210M, for structural pipe and tube. Specification B429/B429M, and for seamless pipe and tube used in pressure applications, Specification B241/B241M.

Nore 3—Structural pipe and tube produced in accordance with this specification is not intended for fluid-carrying applications involving pressure. Refer to either Specification B210M or B241/B241M, as appropriate, for seamless pipe and tube used in fluid-carrying applications involving pressure.

- 1.2 Alloy and temper designations are in accordance with ANSI H35.1M. The equivalent Unified Numbering System alloy designations are those of Table 1 preceded by A9, for example, A91100 for Aluminum 1100 in accordance with Practice E527.
- 1.3 For acceptance criteria for inclusion of new aluminum and aluminum alloys in this specification, see Annex A2.
- 1.4 This specification is the metric counterpart of Specification B221.
- 1.5 The values stated in SI are to be regarded as standard. To other units of measurement are included in this specification.

2. Referenced Documents

2.1 The following documents of the issue in effect on the date of material purchase form a part of this specification to the extent referenced herein:

¹ This specification is under the jurisdiction of ASTM Committee B07 on Light Metals and Alloys and is the direct responsibility of Subcommittee B07.03 on Aluminum Alloy Wrought Products.

Current edition approved Aug. 10, 2007. Published August 2007. Originally approved in 1979. Last previous edition approved in 2005 as B221M - 05n. DOI: 10.1520/30221M-07.

2.2 ASTM Standards;2

B210M Specification for Aluminum and Aluminum-Alloy Drawn Seamless Tubes (Metric)

B211M Specification for Aluminum and Aluminum-Alloy Bar, Rod, and Wire (Metric)

B241/B241M Specification for Aluminum and Aluminum-Alloy Scamless Pipe and Scamless Extruded Tube

B429/B429M Specification for Aluminum-Alloy Extruded Structural Pipe and Tube

B557M Test Methods for Tension Testing Wrought and Cast Aluminum- and Magnesium-Alloy Products (Metric)

B594 Practice for Ultrasonic Inspection of Aluminum-Alloy Wrought Products for Acrospace Applications

B660 Practices for Packaging/Packing of Aluminum and Magnesium Products

B666/B666M Practice for Identification Marking of Aluminum and Magnesium Products

B807/B807M Practice for Extrusion Press Solution Heat Treatment for Aluminum Alloys

B881 Terminology Relating to Aluminum- and Magnesium-Alloy Products

B918 Practice for Heat Treatment of Wrought Aluminum Alloys

B945 Practice for Aluminum Alloy Extrusions Press Cooled from an Elevated Temperature Shaping Process for Production of T1, T2, T5 and T10-Type Tempers

E29 Practice for Using Significant Digits in Test Data to Determine Conformance with Specifications

E34 Test Methods for Chemical Analysis of Aluminum and Aluminum-Base Allovs

B55 Practice for Sampling Wrought Nonferrous Metals and Alloys for Determination of Chemical Composition

E527 Practice for Numbering Metals and Alloys in the Unified Numbering System (UNS)

*A Summary of Changes section appears at the end of this standard.

Copyright © ASTM International, 100 Barr Harbor Drive, PO Box C700, West Conshohocken, PA 19428-2959, United States.

² For referenced ASTM standards, visit the ASTM website, www.astm.org. or contact ASTM Customer Service at service@astm.org. For Annual Book of ASTM Standards volume information, refer to the standard's Document Summary page on the ASTM website.



TABLE 1 Chemical Composition Limits A,B,C

			_	Manga-	Magne-	01	7:		Managallina	Other	Elements ^D	Alumainuma
Alloy	Silicon	Iron	Copper	nese	sium	Chromium	Zinc	Titanium	Vanadium	Each	Total ^E	- Aluminum
1060	0.25	0.35	0.05	0.03	0.03	144	0.05	0.03	0.05	0.03	7.000	99.60 min ^F
1100	0.95	Si + Fe	0.05-0.20	0.05	77-1	***	0.10	***	***	0.05	0.15	99.00 min ^F
2014	0.50-1.2	0.7	3.9-5.0	0.40 - 1.2	0.20 - 0.8	0.10	0.25	0.15 ^G	24	0.05 ^G	0.15	remainder
2024	0.50	0.50	3.8-4.9	0.30-0.9	1.2-1.8	0.10	0.25	0.15 ^a	***	0.05 ^G	0.15	remainder
2219	0.20	0.30	5,8-6.8	0.20-0.40	0.02	747	0.10	0.02-0.10	0.05-0.15	0.05 ^H	0.15 ^H	remalnder
3003	0.6	0.7	0.05-0.20	1.0-1.5	***	***	0.10		***	0.05	0.15	remainder
Alclad 3003	2300	3003	Clad with 70	72 alloy	441-	***	227	419	***	249	440	***
3004	0.30	0.7	0.25	1.0-1.5	0.8-1.3	1884	0.25	***		0.05	0.15	remainder
3102	0.40	0.7	0.10	0.05-0.40	AME .	100	0.30	0.10		0.05	0.15	remainder
5052	0.25	0.40	0.10	0.10	2.2-2.8	0.15-0.35	0.10	127	**1	0.05	0.15	remainder
5083	0.40	0.40	0.10	0.40-1.0	4.0-4.9	0.05-0.25	0.25	0.15	***	0.05	0.15	remainder
5086	0.40	0.50	0.10	0.20-0.7	3.5-4.5	0.05-0.25	0.25	0.15		0.05	0.15	remainder
5154	0.25	0.40	0.10	0.10	3.1-3.9	0.15-0.35	0.20	0.20	÷7.	0.05	0.15	remainder
5454	0.25	0.40	0.10	0.50-1.0	2.4-3.0	0.05-0.20	0.25	0.20	101	0.05	0.15	remainder
5456	0.25	0.40	0.10	0.50-1.0	4.7-5.5	0.05-0.20	0.25	0.20	S.	0.05	0.15	remainder
6005	0.6-0.9	0.35	0.10	0.10	0.40-0.6	0.10	0.10	0.10	147	0.05	0.15	remainder
6005A	0.50-0.9	0.35	0.30	0.50'	0.40-0.7	0.30'	0.20	0.10	177	0.05	0.15	remainder
6060	0.30-0.6	0.10-0.30	0.10	0.10	0.35-0.6	0.5	0.15	0.10	***	0.05	0.15	remainder
6061 ^J	0.40-0.8	0.7	0.15-0.40	0.15	0.8-1.2	0.04-0.35	0.25	0.15	300	0.05	0.15	remainder
6063	0.20-0.6	0.35	0.10	0.10	0.45-0.9	0.10	0.10	0.10	Sec	0.05	0.15	remainder
6066	0.9-1.8	0.50	0.7-1.2	0.6-1.1	0.8-1.4	0.40	0.25	0.20		0.05	0.15	remainder
6070	1.0-1.7	0.50	0.15-0.40	0.40-1.0	0.50-1.2	0.10	0.25	0.15	100	0.05	0.15	remainder
6082	0.7-1.3	0.50	0.10	0.40-1.0	0.6-1.2	0.25	0.20	0.10	447	0.05	0.15	remainder
6105	0.6-1.0	0.35	0.10	0.15	0.45-0.8	0.10	0.10	0.10	, 410	0.05	0.15	remainder
6162	0.40-0.8	0.50	0.20	0.10	0.7-1.1	0.10	0.25	0.10	500	0.05	0.15	remainder
6262	0.40-0.8	0.7	0.15-0.40	0.15	0.8-1.2	0.04-0.14	0.25	0.15	111	0.05 ^K	0.15 ^K	remainder
6351	0.7-1.3	0.50	0.10	0.40-0.8	0.40-0.8	0000	0.20	0.20	121	0.05	0.15	remainder
6463	0.20-0.6	0.15	0.20	0.05	0.45-0.9	30001	0.05	>**	***	0.05	0.15	remainder
6560	0.30-0.7	0.10-0.30	0.05-0.20	0.20	0.20-0.6	0.05	0.15	0.10	224	0.05	0.15	remainder
7005	0.35	0.40	0.10	0.20-0.7	1.0-1.8	0.06-0.20	4,0-5.0	0.01-0.06	***	0.054	0.15 ^L	remainder
7072 ^M		SI + Fe	0.10	0.10	0.10	146	0.8-1.3	***	566			remainder
7075	0.40	0.50	1.2-2.0	0.30	2.1-2.9	0.18-0.28	5.1-6.1	0.20^	850	0.05~	0.15	remainder
7116	0.15	0.30	0.50-1.1	0.05	0.8-1.4		4.2-5.2	0.05	0.05	0.050	0.15	remainder
7129	0.15	0.30	0.50-0.9	0.10	1.3-2.0	0.10	4,2-5.2	0.05	0.05	0.050	0.15	remainder
7178	0.40	0.50	1.6-2.4	0.30	2.4-3.1	0.18-0.28	6.3-7.3	0.20	311	0.05	0.15	remainder

A Limits are in weight percent maximum unless shown as a range, or stated otherwise.

Analysis shall be made for the elements for which limits are shown in this table.

For the purpose of determining conformance to these limits, an observed value or a calculated value obtained from analysis shall be rounded to the nearest unit in

the last right-hand place of the figures used in expressing the specified limit, in accordance with the rounding-off method of Practice E29.

Differs includes listed elements for which no specific limit is shown as well as unlisted metallic elements. The producer may analyze samples for trace elements not specified in the specification. However, such analysis is not required and may not cover all metallic Others elements. Should any analysis by the producer or the purchaser establish that an Others element exceeds the limit of Each or that the aggregate of several Others elements exceeds the limit of Total, the material shall be considered nonconforming.

Other Elements—Total shall be the sum of unspecified metallic elements 0.010 % or more, rounded to the second decimal before determining the sum.

The aluminum content shall be calculated by subtracting from 100.00 % the sum of all metallic elements present in amounts of 0.010 % or more each, rounded to the

second decimal before determining the sum.

^a Upon agreement between the purchaser and the producer or supplier, a Zr + Tl limit of 0.20 % max is permitted. Properties in Specification (Table 2) are not based on the Zirconium and Titanium algorithm.

HZirconium, 0.10-0.25 %. The total for other elements does not include zirconium.

'Manganese plus chromium shall total 0.12-0.50.

In 1965 the requirements for 6062 were combined with those for 6061 by revising the minimum chromium from "0.15 %" to" 0.04 %." This action cancelled alloy 6062.

K Bismuth and lead shall be 0.40-0.7 % each.

^L Zirconium 0.08-0.20 %. The total for other elements does not include zirconium.

Composition of cladding alloy applied during the course of manufacture. Samples from finished tube shall not be required to conform to these limits.

Vupon agreement between the purchaser and the producer or supplier, a Zr + Tl limit of 0.25 % max is permitted. Properties in Specification (Table 2) are not based on the Zirconium and Titanium algorithm.

O Gallium 0.03 % max.

E607 Test Method for Atomic Emission Spectrometric Analysis Aluminum Alloys by the Point to Plane Technique Nitrogen Atmosphere

E716 Practices for Sampling Aluminum and Aluminum Alloys for Spectrochemical Analysis

E1004 Test Method for Determining Electrical Conductivity Using the Electromagnetic (Eddy-Current) Method

E1251 Test Method for Analysis of Aluminum and Aluminum Alloys by Atomic Emission Spectrometry

G47 Test Method for Determining Susceptibility to Stress-

Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

Method of Test for Exfoliation Corrosion Susceptibility in 7XXX Series Copper-Containing Aluminum Alloys (EXCO Test) (G34-72)³

2.3 ANSI Standards:2

³ The applicable edition in the use of this specification is G34-72, which is available in the gray pages of the Annual Book of ASTM Standards, Vol 02.02.

H35.1(M) Alloy and Temper Designation Systems for Alu-

H35.2(M) Dimensional Tolerances for Aluminum Mill **Products**

2.4 ISO Standards:4

ISO 209-1 Wrought Aluminum and Aluminum Alloys-Chemical Composition and Forms of Product

ISO 2107 Aluminum, Magnesium and their Alloys-Temper Designation

ISO 6362-2 Wrought Aluminum and Aluminum Alloy Extruded Rod/Bar, Tube, and Profile-Mechanical Properties

2.5 Federal Standard:5

4th Floor, New York, NY 10036, http://www.ausi.org.

Fed. Std. No. 123 Marking for Shipment (Civil Agencies)

⁴ Available from American National Standards Institute (ANSI), 25 W. 43rd St.,

2.6 Military Standard:5

2.7 AMS Specification:6 AMS 2772 Heat Treatment of Aluminum Alloy Raw Materials

MIL-STD-129 Marking for Shipment and Storage

3. Terminology

3.1 Definitions:

Refer to Terminology B881 for definitions of product terms used in this specification.

3.2 Definitions of Terms Specific to This Standard:

3.2.1 capable of-The term capable of as used in this specification means that the test need not be performed by the producer of the material. However, should subsequent testing by the purchaser establish that the material does not meet these requirements, the material shall be subject to rejection.

TABLE 2 Tensile Property Limits^{A,B}

Temper	Specified Sec Thicknes		Area	a, mm²	Tensile S	trength, MPa	Yield Strength (0.2 % offset), MPa		Elongation, 6%, min	
	over	incl	over	incl	min	max	rnin	max	in 50 mm	in 5 × Diameter (5.65 √A
					Aluminum 1	060				
O H112	all		all		60	95	15		25	00
H112	all		all		60	1506.9	15	(8) 9 (8)	25	22 22
	all		all		177.5	25/8/8	(6.00)	30.00	100	
					Aluminum 1					232
0	all		all		75	105	20		25	
H112 F ^D	all		all		75	25/2	20	1100	25	22
	all		all		999	200	7.7.7	3500		22
					Alloy 201		17.5	15(5)51	* * *	1010/4
0	all		all		2.001.5	205		400		
T4)	all		all		345		240	125	12	10
T4510 ^E }					0.15	7.0	240	10.00	12	10
T42 ^F	all		all		345		***			
T6	f	12.50	ali		415	4.4.4	200	2.55	12	10
'6510 ^E	12.50	18.00	all		440	111	365	10000	7	6
T6511 ^E	18.00	10.0	F-9090	16 000	470		400	77.4	3.1.1	6
	18.00	Acces	16 000	20 000	470	5,558	415	+ + +	\$0.00g	6
Г62 ^F	(18.00	all	436.	415	1.17(1)	400 365	* * *	721	5
	18.00	111	250	16 000	415	4(4)4)	365	V 4-4	7	6
- 5	18.00	1.01	16 000	20 000	415	E 4-3;	365	***		6
D	all		all	97.974	4.4.4	104/6: 104/6:	305	* * *	2.50	5
					Alloy 2024			5(4)3	275	4.1.1.
)	all		ali			240		400		
3]	1500.60	6.30	all		395		290	130	12	10
3510 ^E	6.30	18.00	9.11		415	(478(4))	305	1000	120	111
3511 [€]	18.00	35.00	all		450	006545401	315	81818	12 ⁹	10 ^a
	35.00	166		16 000	485	(6.6.4)	360 ^H		52.5	9
10E	35.00	7.7.7	16 000	20 000	470	200	330'	4.4	* * * * * * * * * * * * * * * * * * * *	9
42 ^F	8.4.4	18.00	all	0.900	395	14.4	260	5.5.5	12	7 10
	18,00	35.00	all		395	77.4	260	E.8(8)		9
	35,00	19000	56000	16 000	395	222	260	Kent.	#38(8) #36(8)	9
່ 81 ໂ	35.00	20/9/90	16 000	20 000	395	204.7	260	1999	* (* A)	7
8510=	1.20	6.30	all		440	WOCK.	385	2006	4	
8511 ^E	6.30	35.00	all		455	38 9 8	400	TANKY.	5	4
D D	35.00	9.49	4 <u>4 (</u> #	20 000	455	0000	400	444		4
	all		all		619(4)	0.010	Wala	X 15 K		(6.4.4)

Available from Standardization Documents Order Desk, DODSSP, Bldg. 4, Section D. 700 Robbins Ave., Philadelphia, PA 19111-5098, http:// ⁶ Available from Society of Automotive Engineers (SAE), 400 Commonwealth www.dodssp.daps.mil. Dr., Warrendale, PA 15096-0001, http://www.sae.org.



TABLE 2 Continued

				TAI	BLE 2 Cor	ntinued				
	Specified Sect Thickness		Area, mm² Tensile Si		Tensile Stre	engih, MPa	ogth, MPa Yield Strength (0.2 % offset), MPa			ation, ^C %, min
Temper	over	incl	over	incl	min	max	min	max	in 50 mm	in 5 × Diameter (5.65 √A)
					Alloy 2219					
)	all		all		1000	220	(4)414	125	12	10
Γ31	1 1	12.50	4.4.4	16 000	290	Ca.14-4	180	7406.9	14	12
「3510 ^E	12.50	80.00		16 000	310	0.00	185	911	933	12
62 ^F	T. com	25.00		16 000	370	17.0	250	18-4.4	6	5
	25.00		060404	20 000	370	0000	250	(4) 6.74	16.63	5
81										
Γ8510 ^E Γ8511 ^E	FEE	80.00	1000	16 000	400	live.	290	1114	6	5
185 [] = 18	all		all							
			uii		AH. 0000	300	900	2011	3.04	
					Alloy 3003					
0	all		all		95	130	35	200 ft ;	25	22
1112 -0	all		all		90	0.0	30	(9.16)83	25	22
	all		all		3341	2474.9	20.60	30436	36991	ψ.ψ. 4
					Alclad Alloy 3	003				
)	all	all	all		90	125	30	191	25	Y Y Y
1112	all				90	374	30 ³	(9/78)	25	23.5
					Alloy 3004					
)	all		all		160	200	60	363637	0404040	90406
D	all		all		p + +	1474041	90.00	4.64	1414.41	4.6.4
					Alloy 3102					
1112 ^K	0.70	1.30	all		75	125	30		25	
	0.70	1,00	un un					1000		.010
					Alloy 5052					
0	all		all		170	240	70	1000	:(6:)(30:)	9.63
		ů.			Alloy 5083	}				
)	A1212	130.00 ^L	ories	20 000	270	350	110	200	14	12
- - - - 1111	110	130.004	3694	20 000	275	7337	165	10.4	12	10
1112	4 4 4	130.00 ^L	244	20 000	270	22.	110	2.55	12	10
D	ali		all		*:*:	15.555	****	7.00	(5-6-6)	5.53
					Alloy 5086	3				
0		130.00 ^L	acia.	20 000	240	315	95	(456)	14	12
H111	177	130.004	749	20 000	250	2.22	145	99.8	12	10
H112	***	130.00 ^L	0.00	20 000	240	14.60	95	14.46	12	10
ED .	all		all		1.00	Tabalah	14.4.4	4.64	567	334
					Alloy 5154	ļ				
)	all		вII		205	285	75	1000	1.9/2	(6)614
1112	all		all		205	(*) (*)	75	0000	(F3) (K	0.00
					Alloy 5454	ı				
0	T/a(k	130.00 ^L	WWW	20 000	215	285	85	1,5000	14	12
H111	***	130.00 ^L		20 000	230	200	130	1,00	12	10
1112	7.57	130.00 ^L		20 000	215	0.00	85	5500	12	10
=0	all		all			(9.8.4)	3.4.6	E30-E3	555	2.54
					Alloy 5456	3				
0	(+(+)-)-(130.00 ⁴	(9/9/4)	20 000	285	365	130	200	14	12
H111	93414	130.00 ^L	543667	20 000	290	1000	180	FW 6	12	10
H112	147	130.00 ^L	W W W	20 000	285	120.00	130	200	12	10
=D	all		all			76.66	100	4.00	1 1 1	199
					Alloy 6005	5				
Г1	90.80 M	12.50	all		170	(4.4%)	105	V(V)+1	16	14
Т5	f con	3.20	all		260	0.000	240	A A 9	8	9666
	3.20	25.00	all		260	(9000)	240	654741	10	9
		20,00			Alloy 6005		3.10	7.4-4	.0	- V
T-1			- 0			n	44.			
T1 T5	2.50	6.30 6.30	all		170 260	225	100	50.50	15	701.70
	6.30	25.00	all all		260	2.53	215 215	1000	7 9	8
T61	0,30	6,30	all		260	1955	240	5(5) 5	8	
	4.5.5	9,00	1100		200	0.03	L-10	X E C		(0.000)



TABLE 2 Continued

	Specified Section or Wall Area Thickness, mm			ea, mm² Tensile Strength, MPa				ngth (0.2 %), MPa	Elongation, 6%, mln	
Temper	over	incl	over	Incl	min	max	min	max	in 50 mm	In 5 × Diameter (5.65 √A)
	6.30	25.00	all		260	-0350	240	445	10	9
					Alloy 6060					
T51		3.20	all		150	440	110	18,000	8	(6)604
T61	440	3.20	all		205	7-614	170	.4(4)45	8	(4(4))
	3.20	25.00	all		205	V: 0.74	170	A/A(a)	10	9
					Alloy 6061					
0	all		all			150	15551	110	16	14
T1	**-	16.00	all		180	0.00	95	Same	16	14
[4 [4510 [©]]	all		all		180	. 15	110	30.6	16	14
۲42 ^{<i>F</i>}	all		all		180	(4.90.4	85	14.3092	16	14
T51 T6, T62 ^F)	100	16.00	all		240	190909	205	19.64	8	7
T6510 ^E }	6,30	6.30	all all		260	(4/4)=	240	411	8	* * *
T6511 ^E J	ſ.	550			260	9.9.9	240	4 = 4	10	9
=D	all		all		Pick.	1.4.4	201	Str.	7000	25,02
					Alloy 6063					
Σ Γ1	all	40.50	all		775	130	(#(#0)	\$03.63	18	16
	12.50	12.50 25.00	all all	1000	115	2000	60	(9(4.8)	12	10
T4,T42F	12.50	12.50	all	40400	110 130	1000	55 70	4.4%	14	10
	12.50	25.00	all		125	0.00	60	***		12 12
T5	9.9191	12.50	all		150	777	110	4.11	8	7
°EO	12.50	25.00	atl		145	9.90	105	7.4.5	3.555	7
"52 54	614145	25.00	all		150	205	110	170	8	7
· • T	3.20	3.20 12.50	all all		225 225	2.7.2	205	3004	8	0.00
Γ6, T62 ^κ		3.20	all		205	2005	205 170	IKKS!	10 8	9000
(1)	3.20	25.00	all		205	900E#	170	14.474	10	9
					Alloy 6066					
4 1	ali		all		166.00	200	100	125	16	14
4510 ^E	all		all		275	* 7.5	170	100	14	12
4511 ^E										
[42 ^F	ail		all		275	30.6	165	4.44	14	12
76 76510 [©]	all		all		345	1011	310	541959	8	7
6511E	all									
62 ^r	all		all		345	1111	290	2/2/4	8	7
					Alloy 6070					
6, T62	- 856	80.00	4 4 4	20 000	330	70.7	310	3/8/4	6	5
					Alloy 6082					
6	5.00	20.00	all		310		260	2015	6	θ
6511	20.00 150.00	150.00 200.00	all		310	199	260	***		8
	130,00	200.00	all		280	7404	240			8
1	V7. W-1	10.5	oll	-	Alloy 6105					
5	4 4 4	12.5 3.20	all all		170 250	550	105 240	30.60	16	14
	3.20	25.00	all		250	1 F (C	240	9(4)4	9 10	9
					Alloy 6162					
5, 5510 ^E 5511 ^E	16104	25.00	all		255	200	235.	271	7	6
6, 6510 ^E	200	6,30	all		260	6.406	240	-0.104	8	5.63
6510 ^E	6.30	12.50	all		260	110	240	41414	10	9
					Alloy 6262					

Tensile Strength, MPa

Area, mm²

Yield Strength (0.2 % offset), MPa

Elongation, C %, min

Specified Section or Wall Thickness, mm

Temper	over	inol	over	incl	min	max	min	max	in 50 mm	In 5 × Diameter (5.65 √A)
T6 T6510 [€] T6511 [€]	all		all		260	12.50	240	-0.5	10	9
					Alloy 6351					
T1	111	12.50	, B25A1	13 000	180	29(6)6	90		15	13
T11	23.5	20.00	all		180	100	110	1909/95 5909/95	16	14
T4	113	20.00	all		220	2000	130	19/2/17	16	14
T5	30.00	6.30	ali		260	17/100	240	2.00	8	
	6.30	25.00	all		260	22.5	240	900	10	9
T51	3.20	25.00	11	13 000	250	4.4.4	230	2.53	10	7
T54 T6	11 4 14	12.50		13 000	205	25/4/4	140	0.0545	10	9
10	3.20	3.20 20.00	all all		290 290	25397	255 255	1915/67	8 10	9
	VI20	20.00	Q.17		Alloy 6463	(888)	233	0000	10	9
T1		10.50		40.000						
T5	51515	12.50 12.50	0000	13 000 13 000	115 150	241	60	72.2.2	12	10
T6,T62 ^F	5 4 4	3.20	190900	13 000	205	WARY	110 170	0.00	8	7
	3.20	12.50	(A. A. W.	13 000	205	755	170	9.56	8 10	9
					Alloy 6560			00,000		
T5	2.50	3.20	all		150	100	110		8	
T6	2.50	3.20	all		205	400	170	1997	8	1440 Ma
					Alloy 7005					
T53		20.00	all		345	435	305	13(7)7/	10	9
					Alloy 7116					
T5	3.20	12.50	all		330	1909.5	290		8	7
					Alloy 7129					
T5, T6	31.6	12.50	all		380	777	340	444	9	8
0	- 11				Alloy 7075					
T6)	all	6.00	all		F.10	275	18,4541	165	10	9
T62 ^F	6.30	6.30 12.50	all all		540 560	797974	485	1000	7	3740
T6510 [€]	12.50	70.00	all		560	16.4.1	505 495	(4) 0 (4)	7	6
T6511E	70.00	110.00	90.0 ac	13 000	560	191909	490	12.4	19-419-1	6 6
	70.00	110.004	13 000	20 000	540	244	485	444		5
	110.00	130.00 ^L	0064	20 000	540	4350	470	300	***	5
T73	1.60	6.30	99.4	13 000	470	69.4	400	1.60	7	***
T73510 ^E }	6.30	35.00	6.69	16 000	485		420	14.836	8	7
173511- /	35.00 70.00	70.00 110.00 ⁴	4-9	16 000	475	2111	405	0.678780	3.95(9).95	7
	70.00	110.00	13 000	13 000 20 000	470	-0.0	395	76.000	(6.4)4.1	6
	10.00	1.25	ali	20 000	450 500	(9)(4)-4	380	00.0000	2.80 (0.60)	6
	1.25	3.20	all		510	060904 040409	435 440	0.00	7 7	4.4 %
T76)	3.20	6.30	196969	13 000	510		440	0474140	7	***
T76510 [€] }	6.30	12.50	30.8	13 000	515	4.55	450	4.8.8	7	6
T76511E	12.50	25.00	1.4.6	13 000	515	9.64	450	1111	(414)45	6
	25.00	50.00	100	13 000	515	117.7	450	43430	(8.8.8)	6
	50.00	75.00	444	13 000	510	191919	440	0.00	0.000	6
F ^ρ	75.00	100.00	***	13 000	510	181875	435	28.00 (6)	00.0040	6
	all		all		8000	19.76.4		7474(4)	20000	777
0	all			00.777	Alloy 7178					
•	alt	1.60	all	20 000	565	275	FOE	165	10	9
	1.60	6,30		13 000	565 580	4.4.4	525		(4.634)	3.63
T6)	6.30	35.00		16 000	600	4.9.9	525 540	2.4.4	5	800
T6510 ^E	35.00	60.00	1.04	16 000	595	35.00	530	3.434.4.0	5	4
T6511E	35.00	60.00	16 000	20 000	580	9.00	515	364(4)	4.4.1	4 4
	60.00	80.00	0.00	20 000	565		-10	4 9-91	4 6 6	4



TARLE 2 Continued

	Specified Section or Wall Thickness, mm		Area, mm²		Tensile Strength, MPa		Yield Strength (0.2 % offset), MPa		Elongation, C%, min	
Temper	over	incl	over	inel	min	max	min	max	in 50 mm	in 5 × Diameter (5.65 √A)
	(1.60	all		545	600	505	2.695	972	0.00
	1.60	6,30	444	13 000	565	2.2.2	510		5	4
	6.30	35.00	9000	16 000	595	15(5)3	530	(+:+:+:	5	4
T62 ^F	35.00	60.00	Y 201	16 000	595	0.00	530	(4)4(4)	TOTAL	4
	35.00	60.00	16 000	20 000	580	0.474	515	2939391	25050	4
	60.00	80.00	1000	20 000	565	37.44	490	10000	19.00	4
T76)	(3.20	6.30	11/0	13 000	525	1.11	455	11.0	7	0.00
T76510 [#] }	6.30	12.50	241406	13 000	530	7(11	460	15.55	7	6
T76511€ J	12.50	25.00	***	13 000	530	0.00	460	UT EIA:	120	6
F ^p	all		all		Yes	2000	59(9)93	14100	KKEE	16.00

A The basis for establishment of tensile property limits is shown in Annex A1.

To determine conformance to this specification, each value shall be rounded to the nearest 1 MPa for strength and the nearest 0.5 % for elongation, in accordance with the rounding-off method of Practice E29.

Elongations in 50 mm apply for shapes tested in full section and for sheet-type specimens machined from material up through 12.5 mm in thickness having parallel surfaces. Elongations in $5 \times$ diameter (5.65 \sqrt{A}), where D and A are diameter and cross-sectional area of the specimen respectively, apply to round test specimens machined from thicknesses over 6.30. See 8.1.1 and 8.1.2 for conditions under which measurements are not required.

^D No mechanical properties are specified or guaranteed.

For stress-relieved tempers (73510, T3511, T4510, T4511, T5510, T5511, T6510, T6511, T73510, T73511, T76510, T76511, T8510, T8511), characteristics and properties offer than those specified may differ somewhat from the corresponding characteristics and properties of material in the basic tempers.

Material in the T42 and T62 tempers is not available from the material producers.

^G Minimum elongation for tube, 10 % in 50 mm and 9 % in 5× diameter.

H Minimum yield strength for tube, 330 MPa. Minimum yield strength for tube 315 MPa.

Yield strength is not applicable to tube.

K Only in tube form.

^L Properties not applicable to extruded tube over 70 mm wall thickness.

4. Ordering Information

- 4.1 Orders for material to this specification shall include the following information:
- 4.1.1 This specification designation (which includes the number, the year, and the revision letter, if applicable),
 - 4.1.2 Quantity in pieces of kilograms,
 - 4.1.3 Alloy (Section 7 and Table 1),
 - 4.1.4 Temper (Section 8 and Table 2).
 - 4.1.5 Nominal cross-sectional dimensions as follows:
 - 4.1.5.1 For rod and round wire—diameter,
 - 4.1.5.2 For square-cornered bar and wire—depth and width,
- 4.1.5.3 For sharp-cornered hexagonal or octagonal bar and wire—distance across flats,
- 4.1.5.4 For round tube—outside or inside diameter and wall
- 4.1.5.5 For square or sharp-cornered tube other than round-distance across flats and wall thickness,
- 4.1.5.6 For round-cornered bar, profile, tube other than round, square, rectangular, hexagonal, or octagonal with sharp corners-drawing required, and
 - 4.1.6 Length.
- 4.2 Additionally, orders for material to this specification shall include the following information when required by the purchaser:
- 4.2.1 Whether heat treatment in accordance with Practice B918 is required (9.2),
- 4.2.2 Whether ultrasonic inspection is required (Section 17, Table 3),

TABLE 3 Ultrasonic Discontinuity Limits for Extruded Bar and Drofiles^A

		Thicknes	s, ⁸ mm	Mass max	Max Width,	ma at at
Alloy	-	over	incl	per Piece, kg	Thickness Ratio	Discontinuity Class ^C
2014 2024 2219	}	12.50	200	300	10:1	В
7075	1	12,50	35.00	300	10:1	В
7178	- 1	35,00	200.60	300	10:1	A

A Discontinuities in excess of those listed in this table shall be allowed, subject to the approval of the procuring activity, if it is established that they will be removed by machining or that they are in noncritical areas.

⁸ The thickness of any element of a "profile" shall be deemed to be the smallest

dimension of that element and the discontinuity class applicable to that particular thickness shall apply to that element of the profile.

^c The discontinuity class limits are defined in Section 11 of Practice B594.

- 4.2.3 Whether inspection or witness of inspection and tests by the purchaser's representative is required prior to material shipment (Section 18),
 - 4.2.4 Whether certification is required (Section 22),
- 4.2.5 Whether marking for identification is required (Section 20), and whether marking of lot number for alloys 2014 and 2024 in the T3- and T4-type tempers and alloy 6061 in the T6-type tempers is required (20.2),
- 4.2.6 Whether Practices B660 applies and, if so, the levels of preservation, packaging, and packing required (Section 21.3), and
- 4.2.7 Requirements for tensile property and dimensional tolerance for sizes not specifically covered (8.1.3 and 15.1.1).



- 4.2.8 Whether Titanium and Zirconium algorithm is allowed as shown in Table 1, Footnote G when ordering 2014 or 2024.
- 4.2.9 Whether Titanium and Zirconium algorithm is allowed as shown in Table 1, Footnote N when ordering 7075.

5. Materials and Manufacture

5.1 The products covered by this specification shall be produced by the hot extrusion method or by similar methods at the option of the producer, provided that the resulting products comply with the requirements in this specification.

6. Quality Assurance

- 6.1 Responsibility for Inspection and Tests—Unless otherwise specified in the contract or purchase order, the producer is responsible for the performance of all inspection and test requirements specified herein. The producer may use his own or any other suitable facilities for the performance of the inspection and test requirements specified herein, unless disapproved by the purchaser in the order or at the time of contract signing. The purchaser shall have the right to perform any of the inspections and tests set forth in this specification where such inspections are deemed necessary to ensure that material conforms to prescribed requirements.
- 6.2 Lot Definition—An inspection lot shall be defined as follows:
- 6.2.1 For heat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions traceable to a heat-treat lot or lots, and subjected to inspection at one time.
- 6.2.2 For nonheat-treated tempers, an inspection lot shall consist of an identifiable quantity of material of the same mill form, alloy, temper, and nominal dimensions subjected to inspection at one time.

7. Chemical Composition

7.1 Limits—The material shall conform to the chemical composition limits in Table 1. Conformance shall be determined by analyzing samples taken when the ingots are poured, or samples taken from the finished or semifinished product. If the chemical composition has been determined during the course of manufacture, analysis of the finished product shall not be required.

Note 4—It is standard practice in the United States aluminum industry to determine conformance to the chemical composition limits prior to further processing of ingots into wrought products. Due to the continuous nature of the process, it is not practical to keep a specific ingot analysis identified with a specific quantity of finished material.

- 7.2 Number of Samples:
- 7.2.1 The number of samples taken for determination of chemical composition shall be as follows:
- 7.2.1.1 When samples are taken at the time the ingots are poured, at least one sample shall be taken for each group of ingots poured simultaneously from the same source of molten metal.
- 7.2.1.2 When samples are taken from the finished or semi-finished product, a sample shall be taken to represent each 2000 kg, or fraction thereof, in the lot, except that not more than one sample shall be required per piece.

7.3 Methods of Sampling:

- 7.3.1 Samples for determination of chemical composition shall be taken in accordance with one of the following methods:
- 7.3.1.1 Samples for chemical analysis shall be taken from the material by drilling, sawing, milling, turning, or clipping a representative piece or pieces to obtain a prepared sample of not less than 75 g. Sampling shall be in accordance with Practice E55.
- 7.3.1.2 Sampling for spectrochemical analysis shall be in accordance with Practices E716. Samples for other methods of analysis shall be suitable for the form of material being analyzed and the type of analytical method used.

Note 5—It is difficult to obtain a reliable analysis of each of the components of clad materials using material in its finished state. A reasonably accurate determination of the core composition can be made if the cladding is substantially removed prior to analysis. The cladding composition is more difficult to determine because of the relatively thin layer and because of diffusion of core elements to the cladding. The correctness of cladding alloy used can usually be verified by a combination of metallographic examination and spectrochemical analysis of the surface at several widely separated points.

7.4 Methods of Analysis—The determination of chemical composition shall be made in accordance with suitable chemical (Test Methods E34) or spectrochemical (Test Methods E607 and E1251) methods. Other methods may be used only when no published ASTM test method is available. In case of dispute, the methods of analysis shall be agreed upon between the producer and purchaser.

8. Tensile Properties of Material from the Producer

- 8.1 *Limits*—The material shall conform to the tensile property requirements specified in Table 2.
- 8.1.1 The elongation requirements shall not be applicable to the following:
- 8.1.1.1 Material of such dimensions that a standard test specimen cannot be taken in accordance with Test Methods B557M, and of such profile that it cannot be satisfactorily tested in full section.
 - 8.1.1.2 Material up through 1.60 mm in thickness.
 - 8.1.1.3 Wire up through 3.20 mm in diameter.
- 8.1.2 The measurement for yield strength is not required for wire up through 3.20 mm in diameter.
- 8.1.3 Tensile property limits for sizes not covered in Table 2 shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.
 - 8.2 Number of Specimens:
- 8.2.1 For material having a nominal mass up through 1.7 kg/linear m, one tension test specimen shall be taken for each 500 kg or fraction thereof in the lot.
- 8.2.2 For material having a nominal mass over 1.7 kg/linear m, one tension test specimen shall be taken for each 300 m or fraction thereof in the lot.
- 8.2.3 Other procedures for selecting samples may be employed if agreed upon between the producer or supplier and the purchaser.
- 8.3 Geometry of test specimens and the location in the product from which they are taken shall be as specified in Test Methods B557M.



8.4 Test Methods—The tension tests shall be made in accordance with Test Methods B557M.

8.5 Retests—When there is evidence that the test specimen is defective or is not representative of the lot of material, retesting may be performed in accordance with Section 8 and 9 of Test Methods B557M.

9. Heat Treatment

9.1 Producer and supplier heat treatment for the production of T1, T2, T5, and T10-type tempers shall be in accordance with Practice B945 and for the production of T3, T4, T6, T7, T8, and T9-type tempers, except as noted in 9.3 or otherwise specified in 9.2, shall be in accordance with AMS 2772.

9.2 When specified, heat treatment for the production of T3, T4, T6, T7, T8, and T9-type tempers shall be in accordance with Practice B918.

9.3 Alloys 6005A, 6060, 6061, 6063, 6066, 6070, 6082, 6162, 6262, 6351, 6463, and 6560 may be solution heat-treated and quenched at the extrusion press in accordance with Practice B807/B807M for the production of T3, T4, T6, T7, T8, and T9-type tempers, as applicable.

10. Producer Confirmation of Heat-Treat Response

10.1 In addition to the requirements of Section 8, material in alloys 2014, 2024, and 6061 produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and natural aging for not less than 4 days at room temperature, conform to the properties specified in Table 2 for T42 temper material. The heat-treated samples may be tested prior to 4 days natural aging but if they fail to conform to the T42 temper properties, the test may be repeated after completion of 4 days natural aging without prejudice.

10.2 Alloys 2219, 7075, and 7178 material produced in the O or F temper (within the size limits specified in Table 2) shall, after proper solution heat treatment and precipitation heat treatment, conform to the properties specified in Table 2 for T62 temper material.

10.3 Number of Specimens—The number of specimens from each lot of O temper material and F temper material to be tested to verify conformance with 10.1 and 10.2 shall be as specified in 8.2.

11. Heat Treatment and Reheat-Treatment Capability

11.1 As-received material in the O or F temper in alloys 2014, 2024, and 6061 (within the size limitations specified in Table 2 and without the imposition of cold work) shall be capable of conforming to the properties specified in Table 2 for T42 temper upon being properly solution heat-treated and naturally aged for not less than 4 days at room temperature.

11.2 As-received material in the O and F tempers in alloys 2219, 7075, and 7178 (within the size limitations specified in Table 2 and without the imposition of cold work) shall be capable of conforming to the properties specified in Table 2 for the T62 temper upon being properly solution and precipitation heat-treated.

11.3 Material in alloys and tempers 2014-T4, T4510, T4511, T6, T6510, and T6511, and 2024-T3, T3510, T3511, T81, T8510, and T8511 shall be capable of conforming to the

properties specified in Table 2 for the T42 temper upon being properly resolution heat-treated and naturally aged for not less than 4 days at room temperature.

Nors 6—6061-T4, T6, T4510, T4511, T6510, and T6511 are deleted from 11.3 because experience has shown the reheat-treated material tends to develop large recrystallized grains and may fail to develop the tensile properties shown in Table 2.

11.4 Alloy 2219 in the T31, T3510, T3511, T81, T8510, and T8511 tempers, and alloys 7075 and 7178 in the T6, T651, T6510, and T6511 tempers shall be capable of conforming to the properties specified in Table 2 for the T62 temper upon being properly resolution heat-treated and precipitation heat-treated.

11.5 Material in T3/T31, T3510, T3511, T4, T4510, and T4511 tempers shall be capable of conforming upon being properly precipitation heat-treated, to the properties specified in Table 2 for the T81, T8510, T8511, T6, T6510, and T6511 tempers, respectively.

12. Stress-Corrosion Resistance

12.1 Alloy 7075 in the T73 and T76-type tempers and alloy 7178 in the T76-type tempers shall be capable of exhibiting no evidence of stress-corrosion cracking when subjected to the test specified in 12.2.

12.1.1 For lot-acceptance purposes, resistance to stress-corrosion cracking for each lot of material shall be established by testing the previously selected tension-test samples to the criteria shown in Table 4.

12.1.2 For surveillance purposes, each month the producer shall perform at least one test for stress corrosion resistance in accordance with 12.2 on each applicable alloy temper for each thickness range 20.00 and over produced that month. Each sample shall be taken from material considered acceptable in accordance with the lot-acceptance criteria of Table 4. A minimum of three adjacent replicate specimens shall be taken from each sample and tested. The producer shall maintain records of all lots so tested and make them available for examination at the producer's facility.

12.2 The stress-corrosion cracking test shall be performed on material 20.00 mm and over in thickness as follows:

12.2.1 Specimens shall be stressed in tension in the short transverse direction with respect to grain flow and held at constant strain. The stress level shall be 75 % of the specified minimum yield strength for T73-type tempers and 170 MPa for T76-type tempers.

12.2.2 The stress-corrosion test shall be made in accordance with Test Method G47.

12.2.3 There shall be no visual evidence of stress-corrosion cracking in any specimen, except that the retest provisions of 19.2 shall apply.

13. Exfoliation-Corrosion Resistance

13.1 Alloys 7075 and 7178 in the T76, T76510, and T76511 tempers shall be capable of exhibiting no evidence of exfoliation corrosion equivalent to or in excess of that illustrated by Category B in Fig. 2 of Test for Exfoliation Corrosion Susceptibility in 7xxx Series Copper Containing Aluminum Alloys (EXCO Test) (G34-72) when tested in accordance with 13.1.1.



TABLE 4 Lot Acceptance Criteria for Resistance to Stress Corrosion and Exfoliation Corrosion

		Lot Acceptance Criteria	
Alloy and Temper	Electrical Conductivity, ^A % IACS	Level of Mechanical Properties	Lot Acceptance Status
7075-T73	40.0 or greater	per specified requirements	acceptable
T73510 and 173511	38.0 through 39.9	per specified requirements and yield strength does not exceed minimum by more than 82 MPa	acceptable
	38.0 through 39.9	per specified requirements but yield strength exceeds minimum by more than 82 MPa	unacceptable ⁸
	less than 38.0	any level	unacceptable ⁸
7075-T76, T76510, and	38.0 or greater	per specified requirements	acceptable
T76511	36.0 through 37.9	per specified requirements	unacceptable ⁸
	less than 36.0	any level	unacceptable ^B
7178-T76, T76510, and	38.0 or greater	per specified requirements	acceptable
T76511	35.0 lhrough 37.9	per specified requirements	unacceptable ^a
	less than 35.0	any lovel	unacceptable ^B

A Sampling for electrical conductivity tests shall be the same as for tensile tests as specified in 8.2. Test specimens may be prepared by machining a fiat, smooth surface of sufficient width for proper testing. For small sizes of tubes, a cut-out portion may be flattened and the conductivity determined on the surface. Chemical milling may be used on flat surface samples. The planting conductivity shall be determined in accordance with Practice E1004 in the following localines:

used on flat surface samples. The electrical conductivity shall be determined in accordance with Practice E1004 in the following locations:

[®] When material is found to be unacceptable, it shall be reprocessed (additional precipitation heat treatment or re-solution heat treatment, stress relieving, straightening, and precipitation heat treatment, when applicable).

Section 1	Thickness, mm	
over	through	Location
***	2.50	surface of tension sample
2.50	12.50	subsurface after removal of approximately 10 % of the thickness
12.50	40.00	subsurface at approximate center of section thickness, on a plane parallel to the longitudinal center line of the material.
40.00		subsurface on tension-test specimen surface that is closest to the center of the section thickness and on a plane parallel to the extrusion surface.

13.1.1 For surveillance purposes, each month at least one exfoliation-corrosion test shall be performed for each size range of extrusions produced during that month. The test shall be in accordance with Test for Exfoliation Corrosion Susceptibility in 7xxx Series Copper Containing Aluminum Alloys (EXCO Test) (G34-72) on material considered acceptable in accordance with lot-acceptance criteria of Table 4. Specimens shall be selected at random and shall be, if possible, a minimum of 50 by 100 mm with the 100-mm dimension in a plane parallel to the direction of extrusion. The test location shall be in accordance with that specified in Table 4. The producer shall maintain records of all surveillance test results and make them available for examination at the producer's facility.

13.2 For lot-acceptance purposes, resistance to exfoliation corrosion for each lot of material in the alloys and tempers listed in 13.1 shall be established by testing the previously selected tension-test samples to the criteria shown in Table 4.

14. Cladding

14.1 The aluminum-alloy cladding on clad tube shall comprise the inside surface (only) of the tube and its thickness shall be approximately 10 % of the total wall thickness.

14.2 When the cladding thickness is to be determined on finished tube, transverse cross sections of at least three tubes from the lot shall be polished for examination with a metal-lurgical microscope. Using a 100× magnification, the cladding thickness at four points 90° apart in each sample shall be measured and the average of the 12 measurements shall be taken as the thickness. For a tube having a diameter larger than can be properly mounted for polishing and examination, the

portions of the cross section polished for examination may consist of an arc about 12 mm in length.

15. Dimensional Tolerances

- 15.1 Dimensions—Variations from the specified dimensions for the type of material ordered shall not exceed the permissible variations prescribed in the tables of ANSI H35.2M (see Table 5).
- 15.1.1 Dimensional tolerances for sizes not covered in ANSI H35.2M shall be as agreed upon between the producer and purchaser and shall be so specified in the contract or purchase order.
- 15.2 Sampling for Inspection—Examination for dimensional conformance shall be made to ensure conformance to the tolerance specified.

16. General Quality

16.1 Unless otherwise specified, the extruded bar, rod, wire, profile, and tube shall be supplied in the mill finish and shall be uniform as defined by the requirements of this specification and shall be commercially sound. Any requirement not so covered is subject to negotiation between the producer and purchaser.

16.2 Each bar, rod, wire, profile, or tube shall be examined to determine conformance to this specification with respect to general quality and identification marking. On approval of the purchaser, however, the producer or the supplier may use a system of statistical quality control for such examination.

17. Internal Quality

17.1 When specified by the purchaser at the time of placing the contract or order, each bar or profile over 12.50 mm in



TABLE 5 Tables of ANSI H35.2

lable No.	Tille
11.2	Cross-Sectional Dimension Tolerances: Profiles Except for Profiles in T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
11.3	Diameter or Distance across Flats-Round Wire and Rod – Square, Hexagonal and Octagonal Wire and Bar
11.4	Thickness or Width (Distance Across Flats)-Rectangular Wire and Bar
11.5	Length: Wire, Rod, Bar and Profiles
11.6	Straightness: Rod, Bar and Profiles
11.7	Twist Bar and Profiles
11,8	Flatness (Flat Surfaces)-Bar, Solid Profiles and Semihollow Profiles Except for O, T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
11.9	Flatness (Flat Surfaces)-Hollow Profiles Except for O, T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
11.10	Surface Roughness- Wire, Rod, Bar and Profiles
11.11	Contour (Curved Surfaces) Profiless
11.12	Squareness of Cut Ends- Wire, Rod, Bar and Profiles
11.13	Corner and Fillet Radii- Bar and Profiles
11.14	Angularity- Bar and Profiles Except for O, T3510, T4510, T6510, T73510, T76510, and T8510 Tempers
12.2	Diameter Round Tube Except for T3510, T4510, T6510, T73510, T76510 and T8510 Tempers
12.3	Width and Depth- Square, Rectangular, Hexagonal, Octagonal Tube Except for T3510, T4510, T6510, T73510, T76510 and T8510 Temper
12.4	Wall Thickness- Round Extruded Tube
12.5	Wall Thickness- Other Than Round Extruded Tube
12.6	Length- Extruded Tube
12.7	Twist- Other Than Round Extruded Tube
12.8	Straightness- Tube in Straight Lengths
12.9	Flatness(Flat Surfaces)
12.10	Squareness of Cut Ends
12.11	Corner and Fillet Radii: Tube Other Than Round
12.12	Angularity: Tube Other Than Round
12.13	Surface Roughness: Extruded Tube
12.14	Dents: Extruded Tube

thickness or smallest dimension, in alloys 2014, 2024, 2219, 7075, and 7178 shall be tested ultrasonically in accordance with Practice B594 to the discontinuity acceptance limits of Table 3.

18. Source Inspection

18.1 If the purchaser desires that his representative inspect or witness the inspection and testing of the material prior to hipment, such agreement shall be made by the purchaser and the producer or supplier as part of the purchase contract.

18.2 When such inspection or witness of inspection and testing is agreed upon, the producer or supplier shall afford the purchaser's representative all reasonable facilities to satisfy him that the material meets the requirements of this specification. Inspection and tests shall be conducted so there is no unnecessary interference with the producer's or supplier's operations.

19. Retest and Rejection

19.1 If any material fails to conform to all of the applicable requirements of this specification, it shall be cause for rejection of the inspection lot.

19.2 When there is evidence that a failed specimen was not representative of the inspection lot and when no other sampling

plan is provided or approved by the purchaser through the contract or purchase order, at least two additional specimens shall be selected to replace each test specimen that failed. All specimens so selected for retest shall meet the requirements of the specification or the lot shall be subject to rejection.

19.3 Material in which defects are discovered subsequent to inspection may be rejected.

19.4 If material is rejected by the purchaser, the seller is responsible only for replacement of the material to the purchaser. As much as possible of the rejected material shall be returned to the seller by the purchaser.

20. Identification Marking of Product

20.1 When specified in the contract or purchase order, all material shall be marked in accordance with Practice B666/B666M.

20.2 In addition, alloys 2014, 2024, 2219, 7075, and 7178 in the T6-, T73-, T76-, and T8-type tempers and, when specified, alloys 2014, 2024, and 6061 in the T3- and T4-type tempers and alloy 6061 in the T6-type tempers shall also be marked with the lot number in at least one location on each piece.

20.3 The requirements specified in 20.1 and 20.2 are the minimum; marking systems that involve added information, larger characters, and greater frequencies are acceptable under this specification, and shall be agreed upon by the producer and purchaser.

21. Packaging and Package Marking

21.1 The material shall be packaged to provide adequate protection during normal handling and transportation and each package shall contain only one size, alloy, and temper of material unless otherwise agreed upon. The type of packing and gross mass of containers shall, unless otherwise agreed upon, be at the producer or supplier's discretion, provided they are such as to ensure acceptance by common or other carriers for safe transportation at the lowest rate to the delivery point.

21.2 Each shipping container shall be marked with the purchase order number, material size, specification number, alloy and temper, gross and net masses, and the producer's name or trademark.

21.3 When specified in the contract or purchase order, material shall be preserved, packaged, and packed in accordance with the requirements of Practices B660. The applicable level shall be as specified in the contract or order. Marking for shipment of such material shall be in accordance with Fed. Std. No. 123 for civil agencies and MIL-STD-129 for Military agencies.

22. Certification

22.1 The producer or supplier shall, on request, furnish to the purchaser a certificate stating that each lot has been sampled, tested, and inspected in accordance with this specification and has met the requirements.

23. Keywords

23.1 aluminum alloy; extruded bars; extruded profiles; extruded rods; extruded tubes; extruded wire



ANNEXES

(Mandatory Information)

A1. BASIS FOR INCLUSION OF PROPERTY LIMITS

A1.1 Limits are established at a level at which a statistical evaluation of the data indicates that 99 % of the population obtained from all standard material meets the limit with 95 % confidence. For the products described, mechanical property limits for the respective size ranges are based on the analyses of at least 100 data from standard production material with no more than ten data from a given lot. All tests are performed in accordance with the appropriate ASTM test methods. For

informational purposes, refer to "Statistical Aspects of Mechanical Property Assurance" in the Related Material section of the Annual Book of ASTM Standards, Vol 02.02. Mechanical property limits in this metric issue were derived from the inch-pound system limits that were developed under the above principles. As test data on metric dimensioned specimens are accumulated, some refinement of limits, particularly for elongations measured in 5D, can be anticipated.

A2. ACCEPTANCE CRITERIA FOR INCLUSION OF NEW ALUMINUM AND ALLOYS IN THIS SPECIFICATION

- A2.1 Prior to acceptance for inclusion in this specification, the composition of wrought or cast aluminum or aluminum alloy shall be registered in accordance with ANS1 H35.1(M). The Aluminum Association⁷ holds the Secretariat of ANS1 H35. Committee⁷ and administers the criteria and procedures for registration.
- A2.2 If it is documented that the Aluminum Association could not or would not register a given composition, an alternative procedure and the criteria for acceptance shall be as follows:
- A2.2.1 The designation submitted for inclusion does not utilize the same designation system as described in ANSI H35.1(M). A designation not in conflict with other designation systems or a trade name is acceptable.
- A2.2.2 The aluminum or aluminum alloy has been offered for sale in commercial quantities within the prior twelve months to at least three identifiable users.
- A2.2.3 The complete chemical composition limits are submitted.
- A2.2.4 The composition is, in the judgment of the responsible subcommittee, significantly different from that of any other aluminum or aluminum alloy already in the specification.

- A2.2.5 For codification purposes, an alloying element is any element intentionally added for any purpose other than grain refinement and for which minimum and maximum limits are specified. Unalloyed aluminum contains a minimum of 99.00 % aluminum.
- A2.2.6 Standard limits for alloying elements and impurities are expressed to the following decimal places:

Less than 0.001 % 0.000X
0.001 to but less than 0.01 % 0.000X
0.01 to but less than 0.10 %
Unalloyed aluminum made by a refining process 0.00X
Alloys and unalloyed aluminum not made by a refining pro- 0.00X
cess
0.10 through 0.55 % 0.XX
(It is customary to express limits of 0.30 through 0.55 % as 0.X0 or 0.X5.)
Over 0.55 % 0.X, X.X, and so

(except that combined Si + Fe limits for 99.00 % minimum aluminum must be expressed as 0.XX or 1.XX)

A2.2.7 Standard limits for alloying elements and impurities are expressed in the following sequence: Silicon; Iron; Copper; Manganese; Magnesium; Chromium; Nickel; Zinc (Note A2.1); Titanium; Other Elements, Each; Other Elements, Total; Aluminum (Note A2.2).

Note A2.1—Additional specified elements having limits are inserted in alphabetical order of their chemical symbols between zinc and titanium, or are specified in footnotes.

Note A2.2—Aluminum is specified as *minimum* for unalloyed aluminum and as a *remainder* for aluminum alloys.

⁷ Available from Aluminum Association, Inc., 1525 Wilson Blvd., Suite 600, Artington, VA 22209, http://www.aluminum.org.



APPENDIXES

(Nonmandatory Information)

X1. ISO EQUIVALENTS OF ANSI ALLOYS AND TEMPERS

TABLE X1.1 ISO Equivalents of Alloys in B221M

Alloys								
ANSI	IŝO	ANSI	Iso					
1060	Al 99.6	6005	Al SiMg					
1100	Al 99.0 Cu	6005A	Al SIMg (A)					
2014	Al Cu4SiMg	6060	Al MgSI					
2024	Al Cu4Mg1	6061	Al Mg1SICu					
2219	Al Cu6Mn	6263	Al Mg0.7Si					
3003	Al Mn1Cu	6262	Al Mg1SiPb					
3004	Al Mn1Mg1	6351	Al Si1Mg0.5Mn					
5052	Al Mg2.5	7005	Al Zn4,5Mg1.5Mn					
5083	Al Mg4.5Mn0.7	7075	Al Zn5.5MgCu					
5086	Al Mg4	7178	Al Zn7MgCu					
5154	Al Mg3.5	. 11 0	· ·· Ziii Mgod					
5454	Al Mg3Mn							
5456	Al Mg5Mn1							

TABLE X1.2 ISO Equivalents of Tempers in B221M

	Tempers							
ANSI	ISO							
F	F							
0	Ö							
H112	M							
T1	TA							
T3	TD							
T3 T4 T5	ŤΒ							
T5	TF							
T6	TE TF							
17	тм							
T8	TH							

X1.1 International Organization for Standardization (ISO) equivalents of the ANSI alloys and tempers given in Table X1.1 and Table X1.2 are included in ISO 209-1: 1989, Part 1, Chemical Composition and ISO 2107-1983. Mechanical prop-

erty limits shown in Part 2, Mechanical Properties, of ISO 6362-2: 1990 are similar to B221M but not necessarily identical.

X2. DESIGNATIONS FOR METALS AND ALLOYS FORMERLY ASSIGNED IN CONFORMANCE WITH PRACTICE B275

X2.1 Designations assigned in conformance with this practice were used for wrought aluminum and wrought aluminum alloys in ASTM specifications prior to 1960 and for cast aluminum and aluminum alloys and ingot prior to 1974, but now designations conforming to the American National Standard Alloys and Temper Designation Systems for Aluminum

(ANSI H35.1) are standard with the UNS, Practice E527 for information only. The former ASTM designations and the corresponding ANSI and UNS designations for wrought alloys are as shown in Table X3.1. Cast alloys and ingot are as shown in Table X3.2.

B221M - 07

TABLE X2.1 Wrought Aluminum Alloys

ANSI H35.1	Designations Former B275 – 63	UNS	ANSI H35,1	Designations Former B275 – 63	UNS
1060 1100 2011 2014 2017 2018 2024 2117 3003	996A 990A CB60A CS41A CM41A CN42C CG42A CG30A	A91060 A91100 A92011 A92014 A92017 A92018 A92024 A92117	5056 5083 5086 5154 5254 5454 5456 5652	GM50A GM41A GM40A GR40A GR40B GM31A GM51A GR20B	A95056 A95083 A95086 A95186 A95254 A95254 A95456 A95652
3003 3004 4032 5005 5050 5052	M1A MG11A SG121A G1B G1A GR20A	A93003 A93004 A94032 A95005 A95050	6053 6061 6063 6101 7075	GS11B GS11A GS10A GS10B ZG62A	A96053 A96061 A96063 A96101 A97075

SUMMARY OF CHANGES

Committee B07 has identified the location of selected changes to this standard since the last issue (B221 - 05a) that may impact its use (Approved Nov. 1, 2006).

- (1) Modified Note 2 in 1.1 to resemble similar comment in B429/B429M.
- (2) Added Note 3 to 1.1.
- (3) Added reference to B945 to 2.2.
- (4) Added reference to B945 to 9.1.
- (5) Added applicable tempers to 9.1.

- (6) Added applicable tempers to 9.2.
- (7) Added applicable tempers to 9.3.
- (8) Added 6060 to list alloys in 9.3.
- (9) Added 6060-T61 temper to Table 2.
- (10) Added 6063-T54 temper to Table 2.
- (11) Added Table X2.1 for reference in the Appendix.

Committee B07 has identified the location of selected changes to this standard since the last issue (B221M - 02) that may impact its use. (Approved September 2004.)

- (1) Added a values statement to 1.5.
- (2) Added reference to B881 to 2.2.
- (3) Deleted reference to E227 from 2.2.
- (4) Replaced 3.1.1 to 3.1.8 with reference B881 in Section 3.
- (5) Added reference to Table 1, Footnote G in 4.2.8,
- (6) Added reference to Table 1, Footnote N in 4.2.9.
- (7) Revised footnotes to Table 1 in order to correspond to B221
- (8) Added new footnote G to Table 1.

- (9) Added new footnote N to Table 1.
- (10) Revised Mn limit for 6015 to 0.15 in Table 1 IAW AS&D.
- (11) Updated 2014, 6351, 7005, 7075, and 7178 tempers in Table 2 IAW AS&D.
- (12) Added 6005A-T61 in Table 2.
- (13) Added ordering information references to footnote G in Table 1 in 4.2.8.
- (14) Added ordering information references to footnote N in Table 1 in 4.2.9.

Committee B07 has identified the location of selected changes to this standard since the last issue (B221M - 04) that may impact its use. (Approved June 2005.)

- (1) Changed 6105 Mn limit in Table 1 to reflect Aluminum Standards and Data.
- (2) Added alloys 6005A, 6082, 6066, 6070, and 6262 to reflect

alloys in Table 1 and Table 2 that may be produced in the T6 temper when produced in accordance with Practice B807/B807M.



Committee B07 has identified the location of selected changes to this standard since the last issue (B221M - 05) that may impact its use. (Approved October 2005.)

(1) Added references to Alloys 6082 and 6560 to Table 2.

(3) Updated Table 5 to reflect current tables in ANSI H35.2M.

(2) Added references to Alloys 6082 and 6560 to Table 1.

(4) Added Alloy 6560 to 9.3.

ASTM International takes no position respecting the validity of any patent rights asserted in connection with any item mentioned in this standard. Users of this standard are expressly advised that determination of the validity of any such patent rights, and the risk of intringement of such rights, are entirely their own responsibility.

This standard is subject to revision at any time by the responsible technical committee and must be reviewed every five years and if not revised, either reapproved or withdrawn. Your comments are invited either for revision of this standard or for additional standards and should be addressed to ASTM International Headquarters. Your comments will receive careful consideration at a meeting of the responsible technical committee, which you may attend. If you feel that your comments have not received a fair hearing you should make your views known to the ASTM Committee on Standards, at the address shown below.

This standard is copyrighted by ASTM International, 100 Barr Harbor Drive, PO Box C700, West Conshohocken, PA 19428-2959. United States. Individual reprints (single or multiple copies) of this standard may be obtained by contacting ASTM at the above address or at 610-832-9565 (phone), 610-832-9565 (tax), or service@astm.org (e-mail); or through the ASTM website (www.astm.org). Permission rights to photocopy the standard may also be secured from the ASTM website (www.astm.org).